Abstract #1684



English

Connecting Projects to Complete the In Situ Resource Utilization Paradigm

In Situ resource utilization (ISRU) involves any hardware or operation that harnesses and utilizes 'in-situ' resources to create products and services for robotic and human exploration. Within NASA, ISRU is typically divided into six focus areas: Resource Assessment, Resource Acquisition, Resource Processing/Consumable Production, In Situ Manufacturing, In Situ Construction, and In Situ Energy. Technologies and systems for each focus area are currently being developed in different projects within NASA's Human Exploration and Operations Mission Directorate (HEOMD) and Space Technology Mission Directorate (STMD). This paper will provide an overview of all of the project areas that combine to form the complete ISRU portfolio, including the HEOMD Advanced Exploration Systems (AES) ISRU Technology Development, In Space Manufacturing, and Resource Prospector projects, and the STMD Game Changing Development (GCD) projects in ISRU and Advanced Construction using Mobile Equipment (ACME). HEOMD and STMD joint projects such as the Mars Oxygen In situ Experiment (MOXIE) slated for the Mars 2020 mission will also be included. In addition, projects developing technologies and systems related to ISRU, such as water electrolysis and fuel cells, will be highlighted and discussed. Finally, since ISRU is a capability that involves multiple elements that must be connected to achieve the final objective of enabling a new paradigm in how we explore space, projects developing technologies that will be needed to complete the ISRU mission capabilities, such as mobility, autonomy, and cryogenic fluid management will also be discussed in the paper.

French

No abstract title in French

No French resume

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Profile of Ms. Diane Linne

General

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Biographies

Biography submitted with the abstract

Ms. Diane Linne is a Senior Research Engineer who has worked in Space Propulsion and Exploration at the NASA Glenn Research Center for 31 years. She has a BSE in Aerospace Engineering from the University of Michigan, and an MSE in Aerospace and Mechanical Engineering from Case Western Reserve University. Ms. Linne performs experimental research in rocket propulsion, propellants, and in situ resource utilization. Her research has included ignition and performance of carbon monoxide and oxygen propellants for Mars sample return, and production of propellants on the Moon and Mars. Her research focuses on utilizing new and emerging technologies to continually increase the performance and/or reduce the mass of the total system.

Biography in the user profile

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Connecting Projects to Complete the In Situ Resource Utilization Paradigm

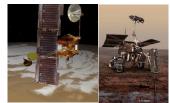
Presented at the Joint Planetary & Terrestrial Mining and Sciences Symposium / Space Resource Roundtable and in conjunction with the Canadian Institute of Mining Convention April 30 – May 2, 2017

Diane L. Linne, NASA/GRC Gerald B. Sanders, NASA/JSC Stanley O. Starr, NASA/KSC



ISRU involves any hardware or operation that harnesses and utilizes 'in-situ' resources to create products and services for robotic and human exploration

Resource Assessment (Prospecting)



Assessment and mapping of physical, mineral, chemical, and water resources, terrain, geology, and environment

In Situ Manufacturing



Production of replacement parts, complex products, machines, and integrated systems from feedstock derived from one or more processed resources **Resource Acquisition**



Excavation, drilling, atmosphere collection, and preparation/ beneficiation before processing

In Situ Construction



Civil engineering, infrastructure emplacement and structure construction using materials produced from *in situ* resources

Radiation shields, landing pads, roads, berms, habitats, etc.

Resource Processing/ Consumable Production



Extraction and processing of resources into products with immediate use or as feedstock for construction & manufacturing

Propellants, life support gases, fuel cell reactants, etc.

In Situ Energy



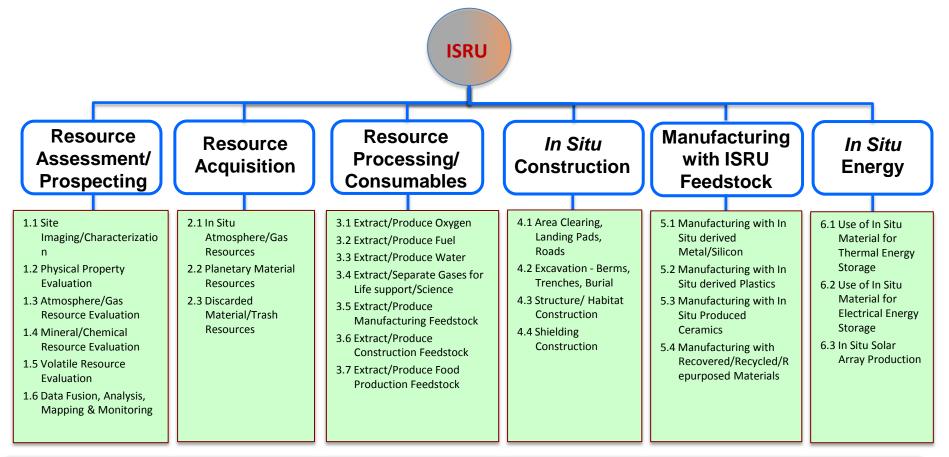
Generation and storage of electrical, thermal, and chemical energy with *in situ* derived materials

Solar arrays, thermal storage and energy, chemical batteries, etc.

- 'ISRU' is a capability involving multiple elements to achieve final products (mobility, product storage and delivery, power, crew and/or robotic maintenance, etc.)
- 'ISRU' does not exist on its own. By definition it must connect and tie to users/customers of ISRU products and services

ISRU Functional Breakdown



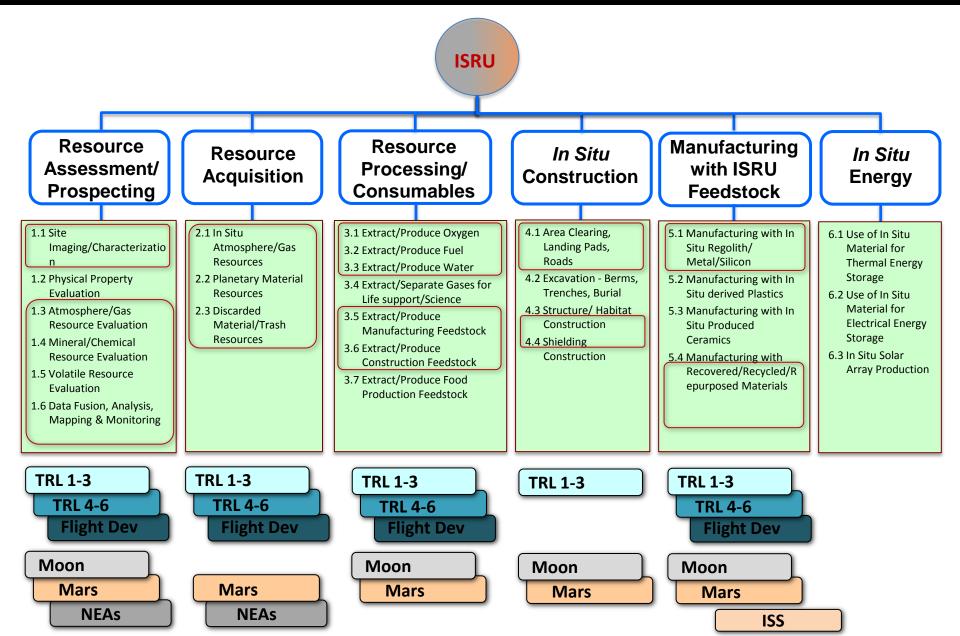


٦	Three Layers of Development: Concept/Technology Feasibility TRL 1-3	L
l	Subsystem/System Dev. in Relevant Environment: TRL 4-6	
	Flight Development	
٦	hree Primary Destinations: Moon Surface	L
l	Mars Surface	

Asteroids/Mars Moons

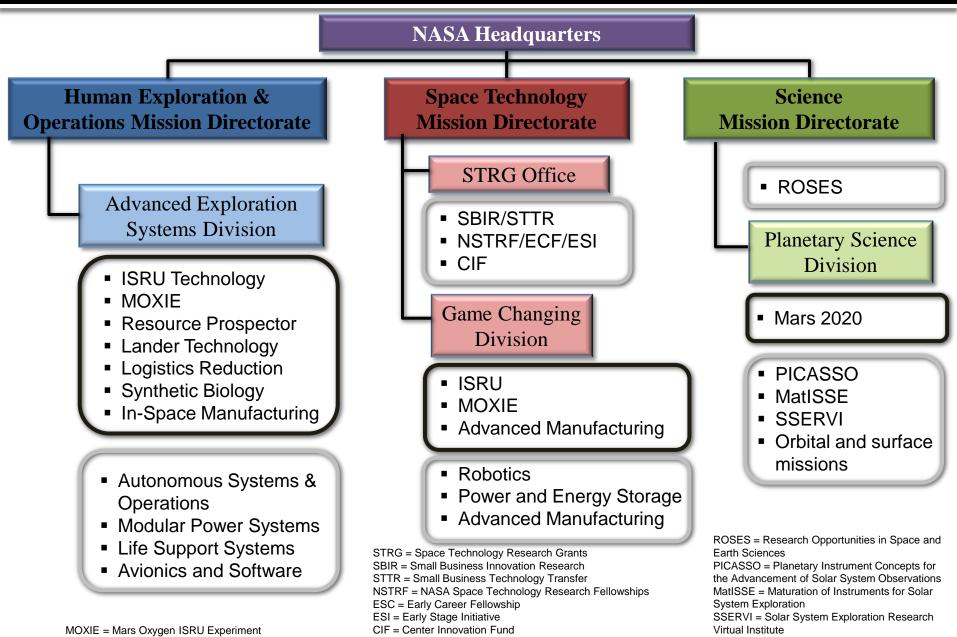
Recent ISRU Related Development within NASA



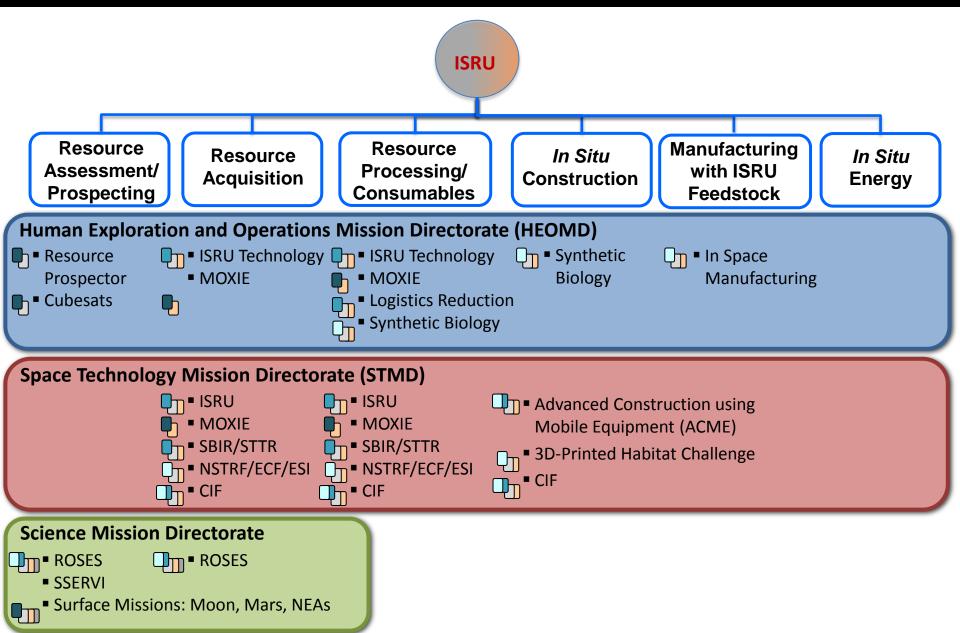


Where Does ISRU Work Reside in NASA?

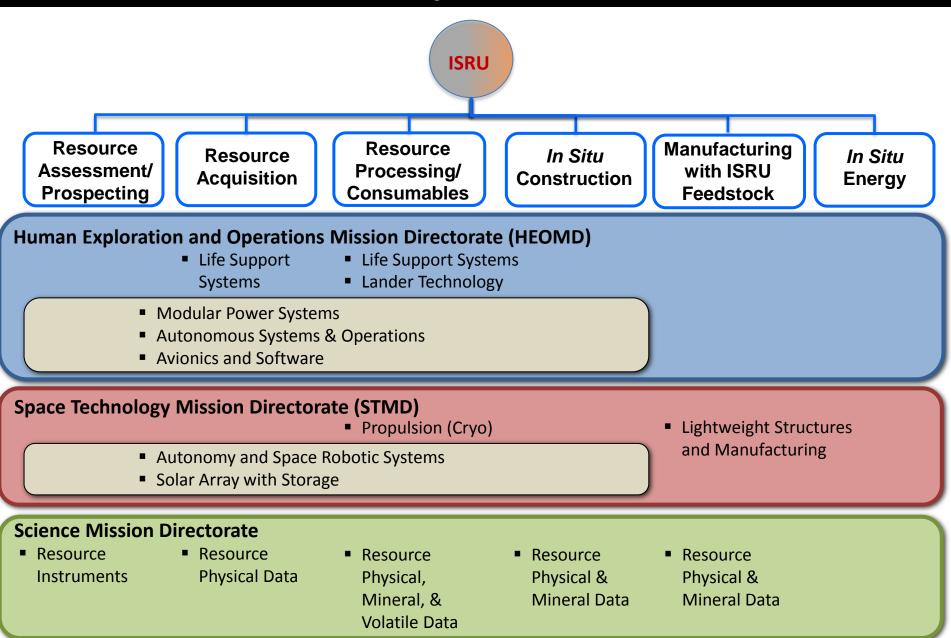




Where Does ISRU-Related Work Reside in NASA? (Projects/Programs)



ISRU Capabilities Requires Information and Hardware from Other Projects



On-Going ISRU Related Work By Project/Program



11: Site Imaging/Characterization X	ISRU Capabilities and Areas of Development	Resource Propsector	AES/STMD ISRU	MOXIE	Synthetic Biology	In Space Manufacturing	ACME	Logistics Reduction	Life Support Systems	ISRU Capabilities and Areas of Development	Resource Propsector	AES/STMD ISRU	MOXIE	Synthetic Biology	In Space Manufacturing	ACME	Logistics Reduction	Life Support Systems
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Resource Prospecting



Resource Assessment (Prospecting) – What Does ISRU Need to Know?



- Terrain
 - Identify specifics such as slope, rockiness, traction parameters
 - Identify what part of ISRU needs each
- Physical / Geotechnical
 - Hardness, density, cohesion, etc.
 - Identify what part of ISRU needs each (e.g., excavation needs to know hardness, density; soil processing needs to know density, cohesion; etc.)
- Mineral
 - Identify specifics
 - Identify what part of ISRU needs each
- Volatile
 - Identify specifics
 - Identify what part of ISRU needs each
- Atmosphere
 - Identify specifics
 - Identify what part of ISRU needs each
- Environment
 - Identify specifics
 - Identify what part of ISRU needs each

Site Characterization and Resource Prospecting on Moon/Mars



Mission	Site & Terrain Properties	Dust Properties	Physical/GeotechnicalSubsurface PropertiesProperties(Indirect Volatiles)		Mineral Characterization	Volatile Characterization
Mars Excursion Rover (MER)	PanCam; Navcam	Magnets	Rock Abrasion Tool (RAT) Microscopic Imager (IM)		Minature Thermal Emission Spec (Mini-TES) Mossbauer Spec (MIMOS II) Alpha particle X-ray spec (APXS)	
Curiosity Rover	Mastcam		Drill/Sieves - Scoop Mars Hand Lens Imager (MAHLI)	Dynamic Neutron Spec (DAN)	ChemCam - LIBS Alpha particle X-ray spec (APXS) X-Ray Diffraction/ Fluorescence (CheMin)	Sample Processing System (SAM) GC/Quadrupole MS Tunable Laser Spec (TLS)
Mars 2020 Rover	Mastcam-Z	Weather/dust measurement (MEDA)		Ground Penetrating Radar (RIMFAX)	X-Ray Fluorescence spec (PIXL) UV Laser-Raman & Luminescence (SHERLOC) SuperCam - LIBS, Raman, Fluorescence, Visible/ IR reflectance	
ExoMars Rover (ESA 2020)	PanCam		Drill (2 m) Close up Imager	Neutron spectrometer Ground Penetrating Radar	IR - mast (1.15-3.3 μm) VIS/IR (0.9-3.5 mm) IR borehole (0.4-2.2 mm) Raman Spectrometer	Sample Processing System GC/MS Laser Desorption-MS
Resource Prospector Rover	360° camera capability on Lander Sterio Camera on Rover		Drill (1 m sample) Measure while drilling Drill Camera	Neutron spectrometer	Near IR	OVEN GC/MS Near IR
Luna 27 (Russia/ESA 2025)	TV imaging	Dust measurements Measurements of plasma/neutrals	Possible arm/scoop Drill (2m) Direct thermal measurement Optical imaging	Seismic measurement Radio measurements of temperature	Neutron/gamma ray spec UV/Optical Imaging IR Spec	Sample Processing System GC/MS and Laser MS

IR = Infrared Spectrometer; VIS = Visiable Light Spectrometer; UV = UltraViolet Spectrometer; MS = Mass Spectrometer; GC = Gas Chromatograph

LIBS = Laser Induced Breakdown Spectroscophy; OVEN = Oxygen and Volatile Extraction Node

Site Characterization and Resource Prospecting on Asteroids/Comets



Mission	Site & Terrain Properties	Dust Properties	Physical/Geotechnical Properties	Subsurface Properties (Indirect Volatiles)	Mineral Characterization	Volatile Characterization
	Cameras		Sampler - pellet impact		X-Ray Fluorescence (XRF)	
Hayabusa	Laser Altimeter (LIDAR)		Thermal sensors on Lander		Near IR	
	Multi-band Imager				Multi-band Imager	
Lander	Camera		Thermal sensors			
	Cameras		Sampler - pellet impact	SCI with Deployable camera	Thermal IR imager	
Hayabusa II	LIDAR		Small Carry-on Impactor (SCI)		Near IR spectrometer	
	Multi-band Imager				Multi-band Imager	
Lander	Multispectral camera	Hyperspectral IR microscope	Radiometer		Multispectral camera	
	Descent imager		Magnetometer		Hyperspectral IR microscope	
Dawn	Framing Camera			Neutron/Gamma Ray spec	Neutron/Gamma Ray spec	
Dawii	Gravity Science-Radio			Sounding radar	Visible/Thermal IR spec	
	Camera- PolyCam	SamCam	Sampler - pneumatic		X-Ray Fluorescence (XRF)	MapCam
OSIRIS-Rex	LIDAR				Visible and IR spectrometer	
					Thermal emission spec	
	Optical imating	Atomic fource microscope	Sounding Radar		Visible/IR thermal spec	lon and neutral analysis MS
Desette		Grain impact analyzer			Optical and IR imager	lon mass analyzer
Rosetta					UV imaging spectrometer	Microwave emission of
						volatiles
Lander	Lander imager	IR and visible analyzer	Harpoon and graplers		Alpha Particle X-Ray spec	SD2
			Sampler, Drill, & Distribution		IR and visible analyzer	GC w/ isotope ratio MS
			(SD2)- down to 23 cm			
			Magnetometer and plasma			
			monitor			

IR = Infrared Spectrometer; VIS = Visiable Light Spectrometer; UV = UltraViolet Spectrometer; MS = Mass Spectrometer; GC = Gas Chromatograph; LIDAR = Light Detection and Ranging

Resource Prospector

Resource Characterization

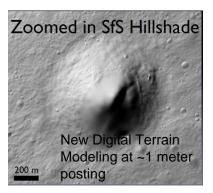
- What: Develop an instrument suite to locate and evaluate the physical, mineral, and volatile resources at the lunar poles
 - Neutron Spectrometer & Near Infrared (IR) to locate subsurface hydrogen/surface water
 - Near IR for mineral identification
 - Auger drill for sample removal down to 1 m
 - Oven with Gas Chromatograph/Mass Spectrometer to quantify volatiles present
- ISRU relevance: Water/volatile resource characterization and subsurface material access/removal

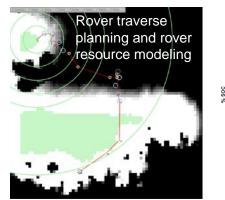
Site Evaluation & Resource Mapping

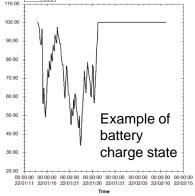
- What: Develop and utilize new data products and tools for evaluating potential exploration sites for selection and overlay mission data to map terrain, environment, and resource information
 - e.g., New techniques applied to generate Digital Elevation Map (DEMs) at native scale of images (~1m/pxl)
- ISRU relevance: Resource mapping and estimation with terrain and environment information is needed for extraction planning
- Mission Planning and Operations
 - What: Develop and utilize tools and procedures for planning mission operations and real time changes
 - Planning tools include detailed engineering models (e.g., power and data) of surface segment systems allows evaluation of designs
 - ISRU relevance: Allows for iterative engineering as a function of environment and hardware performance















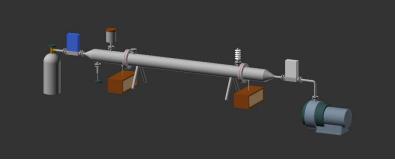
Resource Acquisition

Resource Acquisition – Dust Filtration / Mitigation



Electrostatic precipitator (STMD)

- Assembling components for 2nd generation flow-through precipitator prototype
 - Can vary diameter with three interchangeable tubes (80, 100, 160 mm)
 - Will investigate varying inner electrode diameter (wires to rods) and different electrode materials
- Physics-based model to optimize geometry
 - Modeling equations of motion of particles entering device
- Media filter
 - Physics-based model for scroll media filter
 - Use existing data for validation
 - Mars flow loop, MOXIE
 - Working with MOXIE team for filter analysis and dust loading measurement technique
 - Designing full-scale media filter component for fabrication and testing in FY18



Electrostatic Precipitator Design



Initial set-up of electrostatic precipitator in a flowthrough test



Scroll filter designed for Space Station

Resource Acquisition – CO₂ Compression

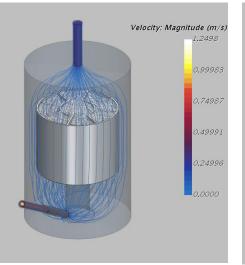


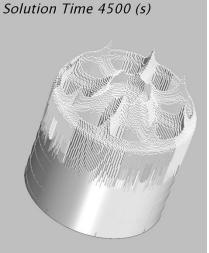
CO₂ Freezer Pump

- Analyzing different cold head designs
 - Finite element modeling of flow and freezing
 - Compare to existing experimental data and iterate
 - Predicted CO₂ solid mass matches experimental results
- Three 'ferris wheel' copper cold heads fabricated for testing

Rapid Cycle Adsorption Pump

- Developing Thermal Desktop / Sinda / Fluint model of microchannel rapid cycle sorption pump
 - Sorbent (Zeolite 13X baseline) is contained in meso-channels
 - Fluid layers for rapid heating/cooling of adsorbent in microchannels
- Addressing modeling / knowledge gaps to simulant Thermal-Swing Adsorption pump
 - Toth and Langmuir 3-site isotherms coded into Sinda / Fluint
 - Adsorption rate, or kinetics, depend mostly on the isotherm
- Design and analysis of realistic system for efficiently cycling temperature of adsorbent in 2 to 6 minute cycles





Gas flow streamlines around cold head

CO₂ solid on cold head



Three 'ferris wheel' copper cold heads for testing; one on right is 3D printed out of GRCop-84

Resource Acquisition – Excavation



Excavation modeling

- Update lunar excavation models to include excavation of different resource types
 - Mars low-water-content loose surface regolith
 - Mars hydrated minerals
 - Icy soils at Moon and Mars
 - Deep ice deposits on Mars
- Validate with existing data and new data when available

Excavator design and architecture

- Use models to evaluate proposed excavation concepts and generate new concepts for mission architecture
- Design, build, and test new and existing excavator concepts and test in relevant environment



Excavation force determination with soil surface 3D measurement using structured light stereography

RASSOR (Regolith Advanced Surface Systems Operations Robot) excavator delivering loose soils





Resource Processing & Consumable Production

Resource Processing / Consumable Production

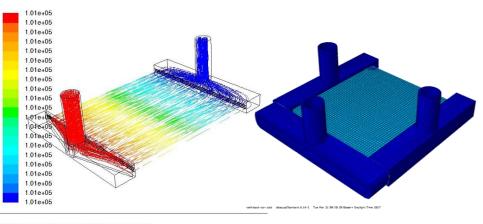


Solid Oxide Electrolysis (SOE) of CO_2

- Baseline SOE stack and insulation model
 - Gathering data for validation and improvements
 - Expanding and reformatting SOE physicsbased performance model
 - Thermal insulation design model
- GRC bi-supported cell fluid & mech. model
 - Evaluate different manifold designs to improve gas distribution through stack
 - Identify stress points caused by thermal loads
 - Recommend design modifications to relieve critical stresses
 - Method will be applied to other SOE designs
- SOE stack scaling limitations
 - Use models to predict limits of active area per cell, # cells per stack

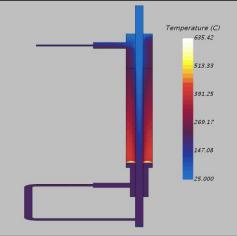
Sabatier Reactor for CH₄ Production

- Sabatier reactor analytical model
- Reviewing state-of-the-art of conventional and microchannel reactor designs
- Catalyst pellets life investigation
 - Analyze new and used catalyst pellets and identify nature of changes over time
 - Guide assessment of longevity/life challenges



Pathlines Colored by Absolute Pressure (pascal) ANSYS Fluent Release 16.1 (3d,

Fluid and mechanical modeling of GRC bi-supported 3-cell stack. (left) pathlines colored by pressure; (right) mechanical stresses



Sabatier reactor thermal CFD model



Thermal camera image of Sabatier reactor during operation. 19

Resource Processing / Consumable Production



Open Reactor Concept

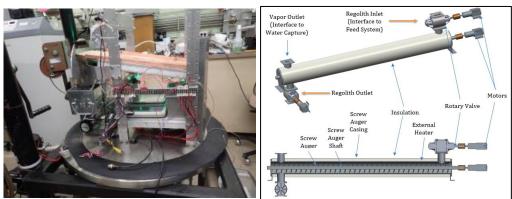
- Open 'air' dryer concept testing completed at GRC
 - Bucket wheel deposits soil on vibrating, heated plate
 - Fan blows Mars atmosphere over plate and sweeps liberated moisture into condenser
- Tested with hydrated mineral, sodium tetraborate decahydrate (Borax), mixed in with GRC-3 simulant
- Physics-based model development

Closed Reactor Concept

- Auger-dryer concept based on terrestrial hardware
 - Physics-based model to assess operation in Mars or lunar environment
- Mars auger-dryer extraction hardware design
 - Hardware to be tested in Mars environment chamber

In-Situ Extraction Concepts

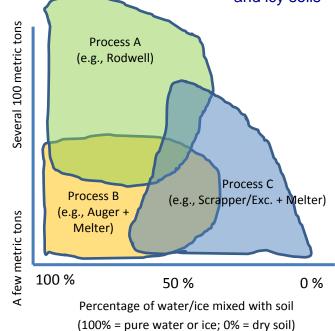
- In-situ extraction modeling
 - Extract the product at the resource location (process raw resource (ice) in place)
 - Working with analytical model developed for "Rodwell" on Earth to determine applicability to Mars (ice/soil mixtures, processing rates)



Open 'air' dryer at NASA GRC

Fotal quantity of water to be withdrawn

Terrestrial auger soil dryer to be modeled for application to Mars and Lunar hydrated and icy soils





In Situ Manufacturing & In Situ Construction

In-Space Manufacturing (AES/GCD ISM Project)

In-Space Manufacturing & Repair Technologies

- What: Work with industry and academia to develop ondemand manufacturing and repair technologies for inspace applications.
 - Two polymer printers currently on ISS' Solicitation for 1st Gen. Multi-material 'FabLab' Rack capable of metallic and electronic manufacturing in-space released
- ISRU relevance: These capabilities can use regolith and other in-situ materials for manufacturing & repair.

In-Space Recycling & Reuse

- What: Develop recycling capabilities to increase mission sustainability.
 - The Refabricator (integrated 3D Printer/Recycler) Tech. Demo. launching to ISS in early 2018.
- ISRU relevance: In-situ materials and products can be recycled for reuse.

In-Space Manufacturing Design Database

- What: ISM is working with Exploration System Designers to develop the ISM database of parts/systems to be manufactured on spaceflight missions.
 - Includes material, verification, and design data. Information will be exported into Utilization Catalogue of parts for crew.
- ISRU relevance: Database to include parts/systems manufactured using in-situ materials.



Additive Manufacturing Facility (AMF) on ISS developed via SBIRs with Made in Space, Inc.



ISS Refabricator (integrated 3D Printer/Recycler) developed via SBIRs with Tethers Unlimited, Inc.



CT Scan (right) of compression cylinder manufactured on ISS (left).



Additive Construction with Mobile Emplacement (ACME) Automated Construction for Expeditionary Structures (ACES)



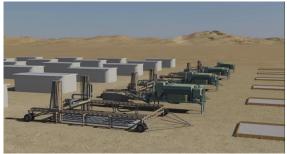
Additive Construction with Mobile Emplacement (ACME) (NASA STMD GCD)

- 2D and 3D printing on a large (structure) scale
 - Use in-situ resources as construction materials to help enable on-location surface exploration
- Demonstrated fabrication of construction material using regolith simulant and multiple binders (polymers, cements)
- Developing zero launch mass (ZLM) print head to extrude a mixture of regolith simulant and high density polyethylene through a heated nozzle
- Use existing NASA GCD robots to position and follow tool paths with regolith print head end effector
- Automated Construction for Expeditionary Structures (ACES) (U.S. Army Corps of Engineers)
 - 3D print large structures to support deployment in remote areas
 - Dry Goods Delivery System provides continuous feedstock from in-situ materials
 - Liquid Goods Delivery System provides continuous flow of liquids/binders
 - Continuous Feedstock Mixing Delivery Subsystem combines all 'ingredients' and performs printing of structure

Standard 2-inch cube compression test specimens



ZLM print head demo illustration



3D print 32' x 16' x 8' barracks with locally sourced concrete, within 48 hrs of deployment



Dry Goods Delivery System



Goal: 3D Print a Habitat for Astronauts using Mars indigenous materials

Prize: \$1.4 million





NASA 360	
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Published on Jan 10, 2017 NASA Centennial Challenges wants YOUR help building a 3-D printed h	abitat on Mare



Synergistic Projects

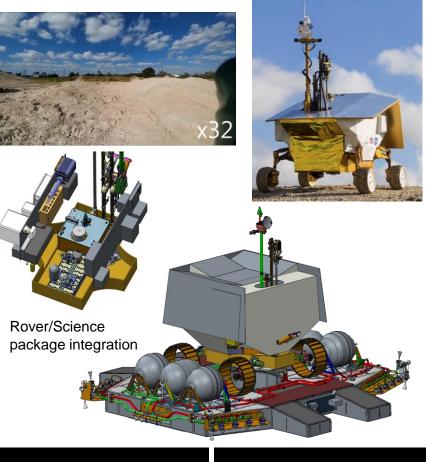
Game Changing Rover Technologies

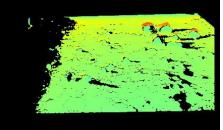
Advanced Mobility

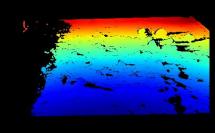
- What: Advanced mobility including active suspension and explicit steering enabling soft soil traversal
 - Active suspension enables terrain traversal
 - Novel wheel, lost cost wheel design
 - Suspension/steering provides rover crawling behaviors
- ISRU relevance: Provides access to lunar permanently shadowed regions for access to volatiles; robust rover mobility in all terrains for prospecting and excavation
- **Resource Prospector Mission integration**
 - What: Developing rover systems that move ISRU payloads around the lunar surface
 - Spectrometers, drills, regolith processing plants
 - ISRU relevance: Rover provides platform for hosting and moving ISRU instruments to target resource area

Rover Lunar Polar Localization and Navigation

- What: Evaluating ability to use stereo for localization and navigation at lunar poles
 - Both low contrast (all gray soil) and high dynamic range (dark shadows and bright sun)
 - Initial results indicate stereo will work at lunar pole
- ISRU relevance: Understanding rover location is vital for prospecting, excavation, and delivery









Autonomy (AES Autonomous Systems and Operations Project)

Autonomous Robotic Operations Planning

- What: Enhance existing tools for use during in-transit, orbital crewed missions
 - Fixed-based kinematics path-planning
- ISRU relevance: Excavation and soil transport

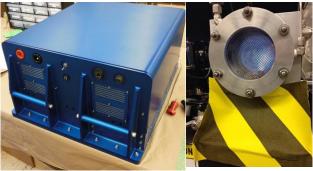
Vehicle Systems Automation

- What: Integrate health management, scheduling and execution across vehicle systems
 - Ties together power and life support operations constraints
- ISRU relevance: ISRU Sabatier and other components of processing plant

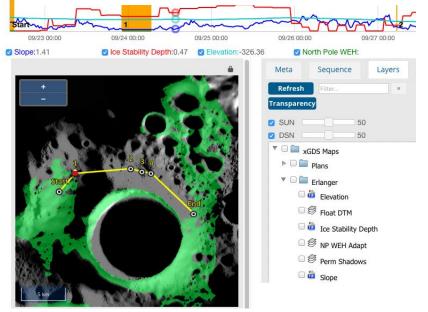
Robotic Mission Planning

- What: Mixed-initiative system that integrates traverse planning and activity planning
 - Planning with temporal, spatial, and spatial-temporal constraints
 - Managing duration uncertainty
- ISRU relevance: excavation and soil transport

Vehicle Systems Automation: testing autonomy components integrated with flight software to operate hardware comparable to that needed for ISRU.



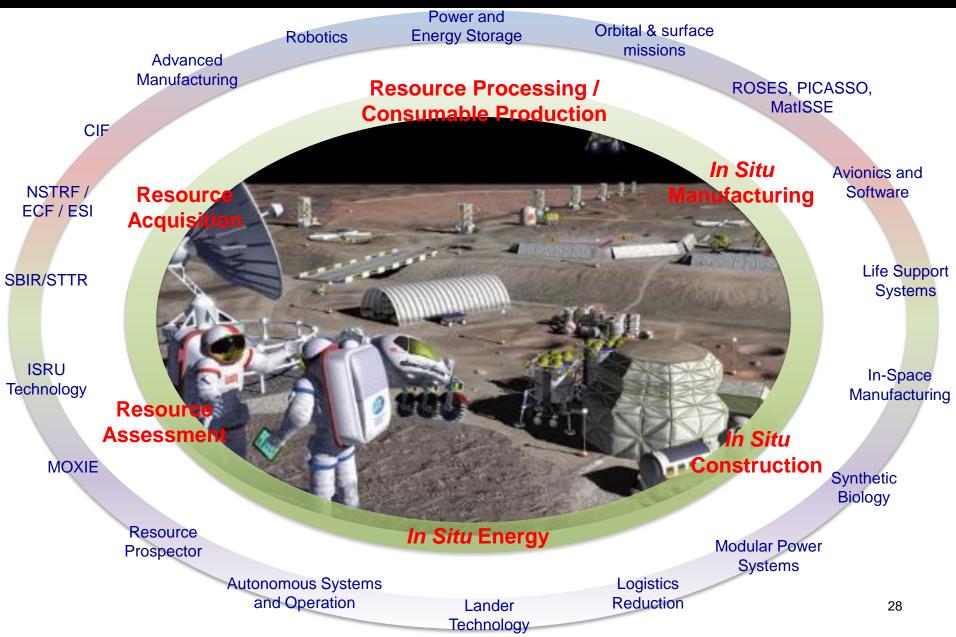
Robotic Mission Planning: Sunlight and communication layers in traverse planner; green areas have communication and dark areas are in shadow





Connecting Projects to Complete the ISRU Paradigm







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Back Up Charts

Current NASA ISRU Missions Under Development



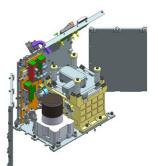


Resource Prospector – RESOLVE Payload

- Measure water (H₂O): Neutron spec, IR spec., GC/MS
- Measure volatiles H₂, CO, CO₂, NH₃, CH₄, H₂S: GC/MS
- Possible mission in 2020

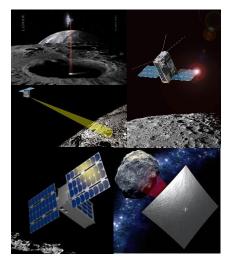
Cubesats (SLS EM-1 2018)

Lunar Flashlight: Uses a Near IR laser and spectrometer to look into shadowed craters for volatiles Lunar IceCube: Carries the Broadband InfraRed Compact High Resolution Explorer Spectrometer (BIRCHES) LunaH-MAP: Carries two neutron spectrometers to produce maps of near-surface hydrogen (H) Skyfire: Uses spectroscopy and thermography for surface characterization NEA Scout: Uses a science-grade multispectral camera to learn about NEA rotation, regional morphology, regolith properties, spectral class



Mars 2020 ISRU Demo

- Make O₂ from Atm. CO₂: ~0.01 kg/hr O₂; 600 to 1000 W-hrs; 15 sols of operation
- Scroll Compressor and Solid Oxide Electrolysis technologies
- Payload on Mars 2020 rover



Space Technology Portfolio



Transformative & Crosscutting Technology Breakthroughs

Pioneering Concepts/Developing Innovation Community

Creating Markets & Growing Innovation Economy

Technology Demonstration

Missions bridges the gap between early proof-of-concept tests and the final infusion of costeffective, revolutionary technologies into successful NASA, government

and commercial space missions.



Small Spacecraft Technology Program

develops and demonstrates new capabilities employing the unique features of small spacecraft for science, exploration and space operations.

Game Changing Development seeks to

identify and rapidly mature innovative/high impact capabilities and technologies that may lead to entirely new approaches for the Agency's broad array of future space missions.



NASA Innovative Advanced

Concepts (NIAC) nurtures visionary ideas that could transform future NASA missions with the creation of breakthroughs-radically better or entirely new aerospace concepts-while engaging America's innovators and entrepreneurs as

partners in the journey.

stimulates and encourages



Space Technology Research Grants seek to

accelerate the development of "push" technologies to support the future space science and exploration needs through innovative efforts with high risk/high payoff while developing the next generation of innovators through grants and fellowships.



Small Business Innovation Research (SBIR) and Small **Business Technology** Transfer (STTR) Programs

Centennial Challenges

directly engages nontraditional

value to NASA's missions and to

the aerospace community. The

program offers challenges set up

as competitions that award prize

money to the individuals or teams

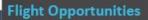
that achieve a specified

technology challenge.

provide an opportunity for small, high technology companies and research institutions to develop key technologies addressing the Agency's needs and developing

the Nation's innovation economy.





facilitates the progress of space technologies toward flight readiness status through testing in space-relevant environments. The program fosters development of the commercial reusable suborbital

transportation industry.



Center Innovation Fund creativity and innovation within the NASA Centers by addressing

the technology needs of the Agency and the Nation. Funds are invested to each NASA Center to support emerging technologies and creative initiatives that leverage Center talent and

capabilities.